PAGE: 1

PRINT DATE: 10/30/89

SMITTLE CRITICAL ITEMS LIST - ORBITER HUMBER: 06-101-0134-X

SUBSYSTEM NAME: ARS - ARPCS

REVISION: 2 10/27/89

PART NAME VEHIOR MANE PART KUMBER VENDOR NUMBER

M2/02 CONTROL PANEL CARLETON TECHNOLOGIES MC250-0002-1001

2720-0001

SRU

LRU

VALVE, TOGGLE

1-4-00-51-45

QUANTITY OF LIKE ITEMS: 2 ONE PER LOOP THO PER SUBSYSTEM

FUNCTION: SUPPLY VALVE, PAYLOAD MANUAL OXYGEN

PROVIDES ON/OFF CONTROL OF 100 PSI OXYGEN TO THE PAYLOAD FROM EITHER CRYO LOOP ONE OR TWO FOR USE IN PAYLOAD OPERATIONS.

PAGE: 2 PRINT DATE: 10/30/89 SHITTLE CRITICAL ITEMS LIST - ORBITER MUMBER: 06-101-0134-02 REVISION 2 10/27/89 SUBSYSTEM: ARS - ARPCS LRU : N2/02 CONTROL PANEL CRITICALITY OF THIS ITEM MAME: VALVE, TOGGLE FAILURE MODE:183 # FAILURE MODE: FAILS OPEN: INTERNAL LEAKAGE MISSION PHASE: PL PRELAUNCH ம LIFT-OFF ŌO T1BS0-NO 80 DE-ORBIT LANDING SAFING ■ YEHICLE/PAYLOAD/KIT EFFECTIVITY: 102 : 103 COLUMBIA DISCOVERY : 104 ATLANTIS 2 . 105 ENDEAVOUR # CAUSE: MECHANICAL SHOCK, VIBRATION, CORROSION, CONTAMINATION, PHYSICAL BINDING/ ■ CRITICALITY 1/1 DURING INTACT ABORT ONLY? X ■ REDUNDANCY SCREEN A) PASS B) FAIL C) PASS PASS/FAIL RATIONALE: **E** A) **m** B)

C)

LINE.

IF PAYLOAD DOES NOT REQUIRE OF AND LINE IS CAPPED AT BULKHEAD, NO PRESSURE MEASUREMENT EXISTS TO INDICATE LEAKAGE INTO PAYLOAD OF SUPPLY

PRINT DATE: 10/30/89

SWITTLE CRITICAL ITEMS LIST - ORBITER MUMBER: 06-1CI-0134-02

- FAILURE EFFECTS -

- (A) SUBSYSTEM: LOSS OF ABILITY TO STOP FLOW TO PAYLOAD O2 SUPPLY LINE.
- (B) INTERFACING SUBSYSTEM(S): NO EFFECT.
- (C) MISSION: NO EFFECT.
- (D) CREW, VEHICLE, AND ELEMENT(S): NO EFFECT.
- E) FUNCTIONAL CRITICALITY EFFECTS:

 INTERNAL LEAKAGE OF THIS VALVE, COMBINED WITH INTERNAL LEAKAGE OF THE

 ASSOCIATED 100 PSI REGULATOR INLET VALVE AND EXTERNAL LEAKAGE IN THE

 PAYLDAD OR SUPPLY LINE, WOULD REQUIRE SHUTGHT OF THE PROBLECS VALVE COULD RESULT

 RESULTING IN LOSS OF DICHALF OF THE REQUIRED LES FLOW AND POSSIBLE LOSS TABLEFICIEN

 OF CREW VEHICLE, OR FLOW TO LESS AND POSSIBLE LOSS OF CREW/VEHICLE IF

- DISPOSITION RATIONALE -

- a (A) DESTER: VALVE BODY IS MADE OF 6061-T6 ALLININUM ANODIZED FOR CORROSION RESISTANCE. FITTINGS ARE MADE OF 17-4 PH CONDITION A CRES, WHICH IS PRECIPITATION HARDENED CORROSION RESISTANT STEEL AND HAS A HIGH STRENGTH TO WEIGHT RATIO. STATIC SEALS ARE MADE OF SILASTIC 675 SILICONE RUBBER. POPPET IS PRESSURE COMPENSATED THROUGH THE USE OF DYNAMIC SEALS AT EACH END, WHICH SLIDE ON THE VALVE STEM. VALVE STEM IS HIGHLY POLISHED FOR EASE OF OPERATION (REDUCED FRICTION PROTECTS SEALS). DYNAMIC SEALS ARE ALSO SILASTIC 675 SILICONE AND ARE LUBRICATED WITH BRAYCO LUBE. SILASTIC 675 SILICONE RUBBER HAS GOOD RESISTANCE TO ENVIRONMENTAL EXPOSURE, FLEXING AND FATIGUE. IT ALSO HAS LOW FLAMMABILITY AND OUTGASSING. THE OZONE RESISTANCE OF SILICONE RUBBER IS EXCELLENT. BRAYCO LUBE IS COMPATIBLE WITH LOW AND HIGH PRESSURE GOZ. INLET/OUTLET PORTS ARE FILTER PROTECTED TO 25 MICRONS. CONSTANT SEAT FORCES DUE TO BELLEVILLE CLOSING SPRING ELIMINATE EXCESS SEAL AND SEAT MEAR. OPERATING FORCE IS 4.5 POUNDS MAXIMUM AND IS INDEPENDENT OF PRESSURE LOADS. THE MOST PROBABLE LEAK (CUT O-RING) IS ESTIMATED AT SO SCON (0.175 LB/HR).
- (8) TEST: ACCEPTANCE TEST - ATP ON VALVE INCLUDES PROOF TEST AT 1875 PSIG (1.5

PRINT DATE: 10/30/89

PAGE: 4

SHITTLE CRITICAL ITEMS LIST - ORBITER MANUER: 06-101-0134-02

TIMES OPERATING PRESSURE). EXTERNAL LEAK TESTED FOR 0.2 SCCM MAX LEAKAGE AT 1250 PSIG FOR A MINIMUM OF 15 MINUTES. INTERNAL LEAK TESTED FOR 2 SCCM MAX LEAKAGE AT 1250 PSIG FOR A MINIMUM OF 15 MINUTES. ATP ON M2/02 CONTROL PANEL AS AN ASSEMBLY INCLUDES EXAMINATION OF PRODUCT, RADIOGRAPHIC INSPECTION, PROOF PRESSURE AT 1870 +/- 20 PSIG. AND EXTERNAL LEAKAGE TEST (DECAY TEST USING GN2) AT 900 +/- 15 PSIG WITH MITROGEN SYSTEM AT A LOWER PRESSURE - ENTIRE PANEL LEAKAGE IS LIMITED TO 11.0 SCCM MAX.

QUALIFICATION TEST - COMPONENT BURST PRESSURE IS 2500 PSIG (2 TIMES OPERATING PRESSURE). SUBJECTED TO THE FOLLOWING AS PART OF THE N2/O2 CONTROL PANEL. RANDOM VIBRATION SPECTRUM - 20 TO 150 NZ INCREASING AT 6 DB/OCTAVE TO 0.03 6**2/HZ AT 150 HZ. CONSTANT AT 0.03 6**2/HZ FROM 150 TO 1000 HZ. DECREASING AT 6 DB/OCTAVE FROM 1000 TO 2000 HZ FOR 48 MINUTES PER AXIS FOR THREE CRTHOGONAL AXES. DESIGN SHOCK - 20G TERMINAL SAWTOOTH PULSE OF 11 MS DURATION IN EACH DIRECTION OF THREE CRTHOGONAL AXES. ATP TO VERIFY LEAKAGE IS PERFORMED AFTER SHOCK AND VIBRATION TESTING.

IN-VEHICLE TESTING - INTERNAL LEAK TEST IS PERFORMED AT 925 - 950 PSIG. 10 SCON NAX LEAKAGE.

OMRSO - 900 AND 100 PSI 02 SYSTEMS 1 AND 2 EXTERNAL LEAK CHECK WAS PERFORMED WITH THIS VALVE CLOSED PRIOR TO FIRST REFLIGHT OF EACH VEHICLE AND WILL BE PERFORMED IF AN LRU IS REPLACED. MAX ALLOWABLE SYSTEM LEAKAGE IS 40 SCCM.

(C) INSPECTION: RECEIVING INSPECTION RAW MATERIAL VERIFIED BY INSPECTION FOR MATERIAL AND PROCESS CERTIFICATION.

CONTAMINATION CONTROL CLEANLINESS LEVEL 200A PER MAG110-301 VERIFIED BY INSPECTION.

ASSEMBLY/INSTALLATION
TORQUES VERIFIED BY INSPECTION. SPRING FORCES VERIFIED BY INSPECTION.
DIMENSIONAL CHECKS PERFORMED BY INSPECTION. MIPS FOR CONCENTRICITY AND
PERPENDICULARITY. 10X VISUAL INSPECTION ON SEAL RING VERIFIED BY
INSPECTION.

NONDESTRUCTIVE EVALUATION INSPECTION OF WELDS BY ZOX VISUAL EXAM, X-RAY AND PENETRANT.

CRITICAL PROCESSES

PAGE: 3 PRINT DATE: 10/30/89

SHITTLE CRITICAL ITEMS LIST - ORBITER MIMBER: 06-101-0134-02

PARTS PASSIVATION AND ANODIZING VERIFIED BY INSPECTION. HEAT TREATMENT VERIFIED BY INSPECTION. SOLDER CONNECTIONS VERIFIED BY INSPECTION TO BE PER MHB630C.4(3A). POTTING VISUALLY VERIFIED BY INSPECTION. APPLICATION OF LUBRICANT ON SEAL RING VERIFIED BY INSPECTION. TIG WELD SCHEDULE VERIFIED BY INSPECTION.

TESTING ATP VERIFIED BY INSPECTION.

HANDLING/PACKAGING. STORAGE AND SHIPPING PROCEDURES ARE VERIFIED.

(D) FAILURE HISTORY: NO FAILURE HISTORY APPLICABLE TO INTERNAL LEAKAGE FAILURE MODE. THE TOGGLE VALVE HAS SUCCESSFULLY BEEN USED THROUGH THE SHUTTLE PROGRAM CONSIDERING THIS FAILURE MODE.

	OPERATIONAL	USE:	
TB\$.			

- APPROVALS -

RELIABILITY ENGINEERING: D. R. RISING

DESIGN ENGINEERING : K. KELLY
QUALITY ENGINEERING : M. SAVALA

MASA RELIABILITY : MASA SUBSYSTEM MANAGER : NASA QUALITY ASSURANCE :

•

ير مهر

A. Hotter

11/1/69